



AEROSPACE MATERIAL SPECIFICATION

AMS4023™**REV. F**

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Noncurrent	1984-04
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Stabilized	2015-10

Superseding AMS4023E

Aluminum Alloy Sheet and Plate, Alclad
1.0Mg - 0.60Si - 0.28Cu - 0.20Cr (Alclad 6061; -T6 Sheet, -T651 Plate)

RATIONALE

AMS4023F stabilizes this document because it contains mature technology that is not expected to change and thus no further revisions are anticipated.

STABILIZED NOTICE

AMS4023 has been declared "STABILIZED" by the SAE AMS D Nonferrous Alloys Committee. This document will no longer be updated and may no longer represent standard industry practice. This document was stabilized because this document contains mature technology that is not expected to change and thus no further revisions are anticipated.. Previously this document was reaffirmed Noncurrent in 2004. The last technical update of this document occurred in January 1977. Users of this document should refer to the cognizant engineering organization for disposition of any issues with reports/certifications to this specification; including exceptions listed on the certification. NOTE: In many cases, the purchaser may represent a sub tier supplier and not the cognizant engineering organization.

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1. SCOPE:

1.1 Form: This specification covers an aluminum alloy in the form of sheet and plate.

1.2 Application: Primarily for formed, medium-strength, structural parts not requiring heat treatment and which are required to exhibit maximum corrosion resistance and to approximate the color and appearance of other clad aluminum alloy parts.

2. APPLICABLE DOCUMENTS: The following publications form a part of this specification to the extent specified herein. The latest issue of Aerospace Material Specifications (AMS) shall apply. The applicable issue of other documents shall be as specified in AMS 2350.

2.1 SAE Publications: Available from Society of Automotive Engineers, Inc., 400 Commonwealth Drive, Warrendale, PA 15096.

2.1.1 Aerospace Material Specifications:

AMS 2202 - Tolerances, Aluminum-Base and Magnesium-Base Alloy Sheet and Plate

AMS 2350 - Standards and Test Methods

AMS 2355 - Quality Assurance Sampling and Testing of Aluminum-Base and Magnesium-Base Alloys, Wrought Products (Except Forgings and Forging Stock) and Flash Welded Rings

2.2 Government Publications: Available from Commanding Officer, Naval Publications and Forms Center, 5801 Tabor Avenue, Philadelphia, PA 19120.

2.2.1 Military Specifications:

MIL-H-6088 - Heat Treatment of Aluminum Alloys

2.2.2 Military Standards:

MIL-STD-649 - Aluminum and Magnesium Products, Preparation for Shipment and Storage

3. TECHNICAL REQUIREMENTS:

3.1 Composition: Shall conform to the following percentages by weight, determined in accordance with AMS 2355:

Core (6061)			Cladding (7072)		
	min	max		min	max
Magnesium	0.8	1.2	Zinc	0.8	1.3
Silicon	0.40	0.8	Silicon + Iron	--	0.7
Copper	0.15	0.40	Magnesium	--	0.10
Chromium	0.04	0.35	Copper	--	0.10
Iron	--	0.7	Manganese	--	0.10
Zinc	--	0.25	Other Impurities, each	--	0.05
Manganese	--	0.15	Other Impurities, total	--	0.15
Titanium	--	0.15	Aluminum		remainder
Other Impurities, each	--	0.05			
Other Impurities, total	--	0.15			
Aluminum		remainder			

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Silicon	0.40	- 0.8	Silicon + Iron	--	0.7
Copper	0.15	- 0.40	Magnesium	--	0.10
Chromium	0.04	- 0.35	Copper	--	0.10
Iron	--	0.7	Manganese	--	0.10
Zinc	--	0.25	Other Impurities, each	--	0.05
Manganese	--	0.15	Other Impurities, total	--	0.15
Titanium	--	0.15	Aluminum	remainder	
Other Impurities, each	--	0.05			
Other Impurities, total	--	0.15			
Aluminum	remainder				

- 3.2 **Condition:**

3.2.1 **Sheet:** Solution and precipitation heat treated in accordance with MIL-H-6088.

3.2.2 **Plate:** Solution heat treated, stretched to produce a nominal permanent set of 2% but not less than 1-1/2% nor more than 3%, and precipitation heat treated. Heat treatments shall be performed in accordance with MIL-H-6088.

3.2.2.1 Plate shall receive no further straightening operations after stretching.

- 3.3 **Properties:** The product shall conform to the following requirements, determined in accordance with AMS 2355:

3.3.1 **Tensile Properties:** Shall be as specified in Table I and 3.3.1.1.

TABLE I

Nominal Thickness Inches	Tensile Strength psi, min	Yield Strength at 0.2% Offset psi, min	Elongation in 2 in. or 4D %, min
0.010 to 0.020, incl	38,000	32,000	8
Over 0.020 to 0.499, incl	38,000	32,000	10
Over 0.499 to 1.000, incl	42,000	35,000	9
Over 1.000 to 2.000, incl	42,000	35,000	8
Over 2.000 to 4.000, incl	42,000	35,000	6
Over 4.000 to 5.000, incl	40,000	35,000	6

TABLE I (SI)

Nominal Thickness Millimetres	Tensile Strength MPa, min	Yield Strength at 0.2% Offset MPa, min	Elongation in 50.8 mm or 4D %, min
0.25 to 0.51, incl	262	221	8
Over 0.51 to 12.67, incl	262	221	10
Over 12.67 to 25.40, incl	290	241	9
Over 25.40 to 50.80, incl	290	241	8
Over 50.80 to 101.60, incl	290	241	6
Over 101.60 to 127.00, incl	276	241	6

3.3.1.1 Tensile properties of plate over 5.000 in. (127.00 mm) in nominal thickness shall be as agreed upon by purchaser and vendor.

3.3.2 **Bending:** Product 0.010 - 0.499 in. (0.25 - 12.67 mm), incl, in nominal thickness shall withstand, without cracking, bending at room temperature through an angle of 180 deg (3.14 rad) around a diameter equal to the bend factor times the nominal thickness of the product with axis of bend parallel to the direction of rolling.

Nominal Thickness		Bend Factor
Inch	(Millimetres)	
0.010 to 0.036, incl	(0.25 to 0.91, incl)	3
Over 0.036 to 0.064, incl	(Over 0.91 to 1.63, incl)	4
Over 0.064 to 0.128, incl	(Over 1.63 to 3.25, incl)	5
Over 0.128 to 0.249, incl	(Over 3.25 to 6.32, incl)	6
Over 0.249 to 0.499, incl	(Over 6.32 to 12.67, incl)	10

3.3.2.1 Bending requirements for plate over 0.499 in. (12.67 mm) in nominal thickness shall be as agreed upon by purchaser and vendor.

3.3.3 **Cladding Thickness:** After rolling, the average cladding thickness per side shall be not less than 4% of the total composite thickness.

3.4 **Quality:** The product, as received by the purchaser, shall be uniform in quality and condition, sound, and free from foreign materials and from internal and external imperfections detrimental to usage of the product.

3.5 **Tolerances:** Unless otherwise specified, tolerances shall conform to all applicable requirements of AMS 2202.

4. **QUALITY ASSURANCE PROVISIONS:**

4.1 **Responsibility for Inspection:** The vendor of the product shall supply all samples and shall be responsible for performing all required tests. Results of such tests shall be reported to the purchaser as required by 4.4. Purchaser reserves the right to perform such confirmatory testing as he deems necessary to ensure that the product conforms to the requirements of this specification.

4.2 **Classification of Tests:**

4.2.1 **Acceptance Tests:** Tests to determine conformance to composition (3.1), tensile properties (3.3.1), and tolerance (3.5) requirements are classified as acceptance tests.

4.2.2 **Periodic Tests:** Tests to determine conformance to bending (3.3.2) and cladding thickness (3.3.3) requirements are classified as periodic tests.

4.3 **Sampling:** Shall be in accordance with AMS 2355. Frequency of sampling for periodic tests shall be as agreed upon by purchaser and vendor.

4.4 **Reports:**

4.4.1 The vendor of the product shall furnish with each shipment three copies of a report stating that the product conforms to the chemical composition and other technical requirements of this specification. This report shall include the purchase order number, material specification number and its revision letter, size, and quantity.